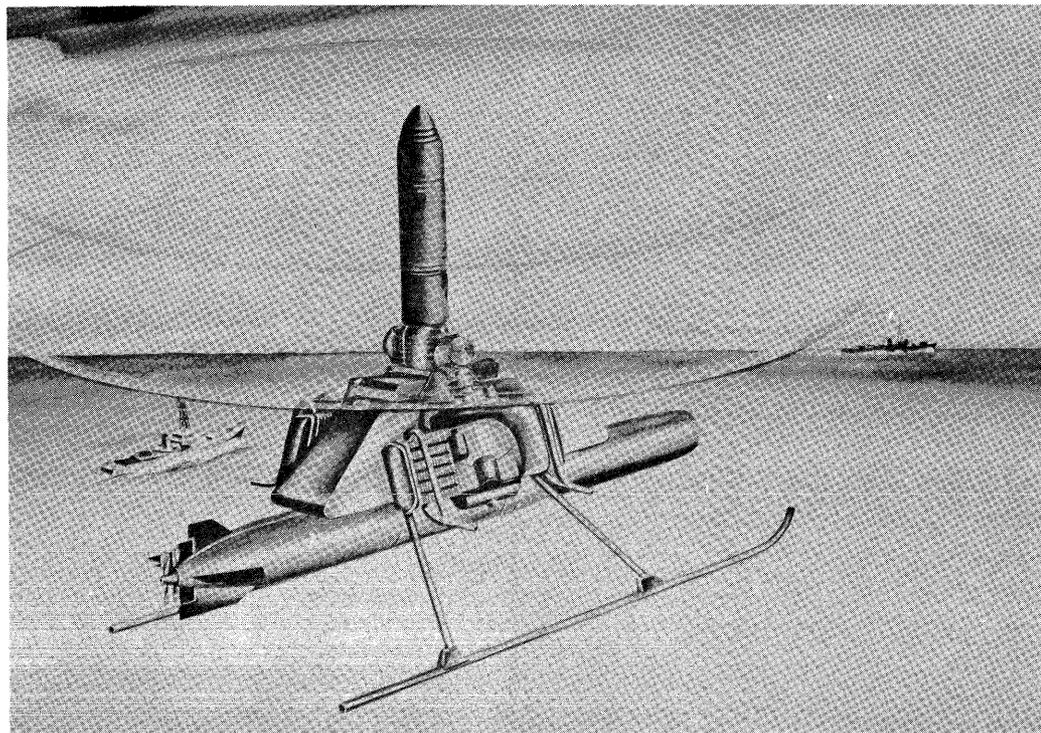


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BY TG Holt ON 9 May 97

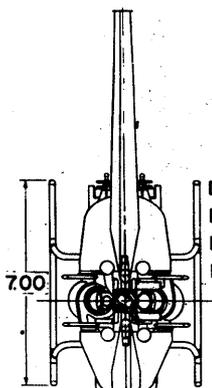


# STANDARD AIRCRAFT CHARACTERISTICS

DSN-2

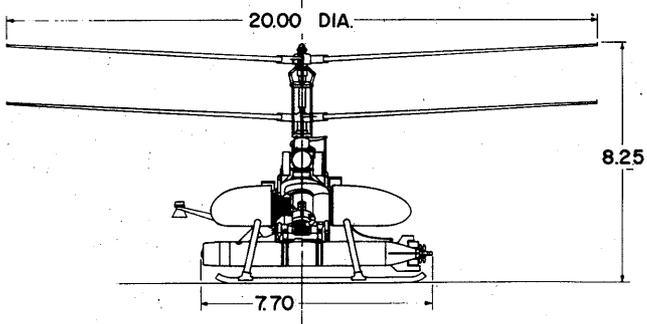
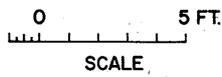
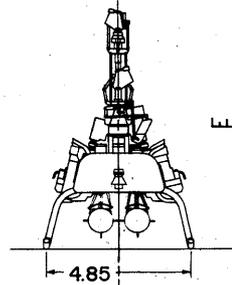
GYRODYNE

Standard Aircraft Characteristics NAVAR 1335B (Rev. 1-55)



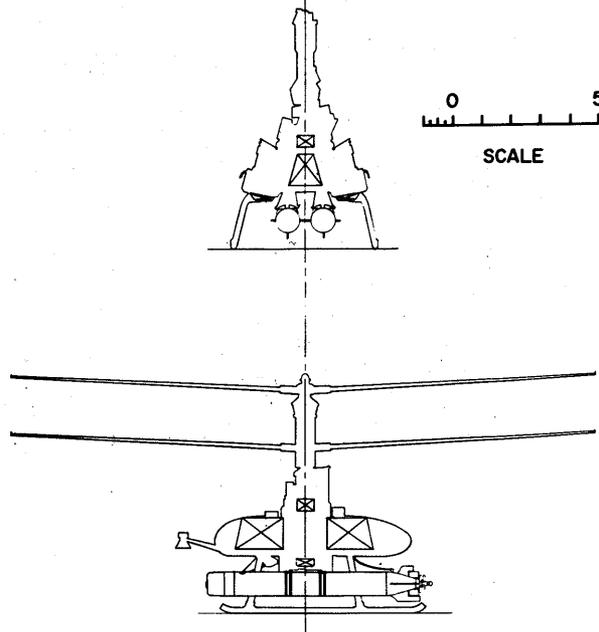
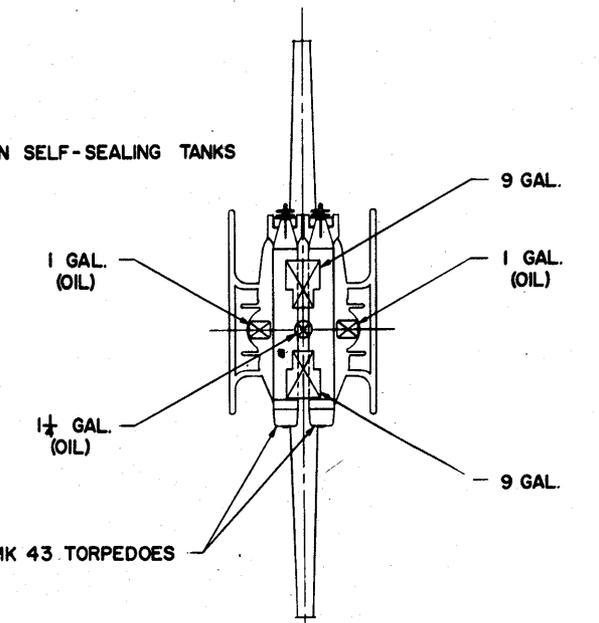
**ROTOR DATA**

DISC AREA - 314.2 SQ. FT.  
 BLADE AREA - 22.6 SQ. FT.  
 BLADE SECTION - NASA 0021 TO 0012  
 ROTOR GEAR RATIO - 8.20 TO 1



DESCRIPTIVE ARRANGEMENT

☒ NON SELF-SEALING TANKS



ARMAMENT & TANKAGE

**POWER PLANT**

ENGINE ..... (2) YO-95-4  
 MFR ..... Gyrodyne-Porsche  
 TYPE ..... 4 cyl., 4 cycle

**RATINGS**

	<u>BHP</u>	<u>RPM</u>	<u>ALT</u>
T.O.	72	4500	SSL
NORM.	72	4500	SSL

Model Spec. No. 2 of 28 Jan  
 1959, revised 5 Aug 1959

**ORDNANCE**Torpedoes:

500 lb. .... 1  
 or  
 MK 43 ..... 2

Max. Load Capacity  
 - 530 lbs.

**MISSION AND DESCRIPTION**

The Model DSN-2 Drone is a remotely controlled ASW Helicopter, designed to carry a 500 lb. weapon (torpedo-type) and be capable of launching this weapon as directed by remote control. An alternate design loading (overload condition) is to carry two MK43 Torpedoes, each weighing 266 lbs.

The drone incorporates two two-bladed coaxial rotors of the semi-rigid (see-saw) type. The blades are of laminated wood construction incorporating taper in planform and thickness and 12° negative twist. Attitude is completely controllable through the rotors by conventional cyclicpitch control; yaw control is achieved by means of rotor blade tip air (drag) brakes.

Two Gyrodyne-Porsche YO-95-4 four cylinder, four cycle reciprocating engines, rated at 72 hp (after cooling) at 4500 rpm at sea level, are incorporated in the DSN-2. Engine cooling is provided by a fan.

Service use of this drone weapons system is not presently contemplated, although it will be used for purposes of testing and checking out the avionics equipment and remote control system to be incorporated in the DSN-3.

**DEVELOPMENT**

First Flight ..... July 1960

**DIMENSIONS**

DISC AREA ..... 314.2 sq. ft.  
 BLADE AREA ..... 22.6 sq. ft.  
 ROTOR DIAMETER ..... 20' - 0"  
 LENGTH ..... 20' - 0"  
 HEIGHT ..... 8' - 3"  
 TREAD ..... 4' - 10"

**WEIGHTS**

<u>LOADINGS</u>	<u>LEB.</u>
EMPTY .....	813
NORMAL .....	1450
OVERLOAD .....	1500

All weights are estimated

**FUEL AND OIL**

<u>NO. TANKS</u>	<u>GALE.</u>	<u>LOCATTON</u>
2	18	Fuselage
GRADE .....	96/96	
SPEC .....	MIL-F-5572B	

**OIL**

CAPACITY .....	3.25 gals
GRADE .....	Symbol 9110/9250
SPEC .....	MIL-L-9000D

**ELECTRONICS**

Gyrodyne-Lear automatic stabilization and remote control equipment

(Nomenclature not yet available)

## PERFORMANCE SUMMARY

TAKE-OFF LOADING CONDITION		(1) NORMAL 1-500# Tor- pedo	(2) OVERLOAD 2-MK 43		
TAKE-OFF WEIGHT	lb.	1450	1500		
Fuel	lb.	95	95		
Payload	lb.	500	530		
Disc loading	lb./sq.ft.	4.62	4.78		
Vertical rate of climb at S.L. (A)	fpm.	640	420		
Absolute hovering ceiling (A)	ft.	2950	1850		
Max. rate of climb at S.L. (A)	fpm.	1160	1040		
Service ceiling (100 fpm) (A)	ft.	8600	7600		
Speed at S.L. (A)	kn.	76.8	76.0		
Max. speed/altitude (A)	kn./ft.	76.8/S.L.	76.0/S.L.		
Max. range	n.mi.	85.3	--		
Average cruising speed	kn.	65	--		
Cruising altitude	ft.	S.L.	--		
Combat radius	n.mi.	32.0	31.2		
Average cruising speed	kn.	77.5	76.8		
Cruising altitude	ft.	S.L.	S.L.		
Max. endurance	hrs.	--	1.60		
Average cruising speed	kn.	--	42		
Cruising altitude	ft.	--	S.L.		

## NOTES

(A) Normal Rated Power ( Take-off Power)

PERFORMANCE is based on calculations modified by contractor flight data of the Model XRON-1 Rotorcycle (17' rotor dia.)

COMBAT RADIUS, RANGE, and ENDURANCE are based on engine specific fuel consumption increased 5%.

All performance is out of ground effect.

COMBAT RADIUS MISSION

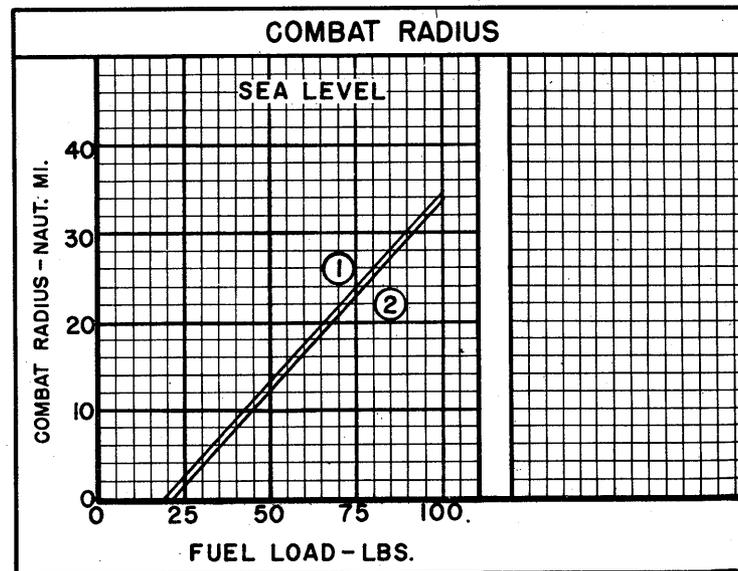
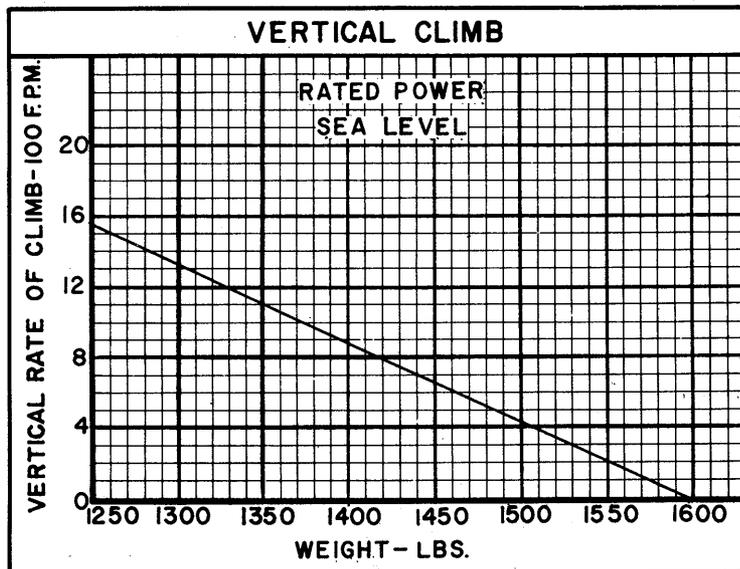
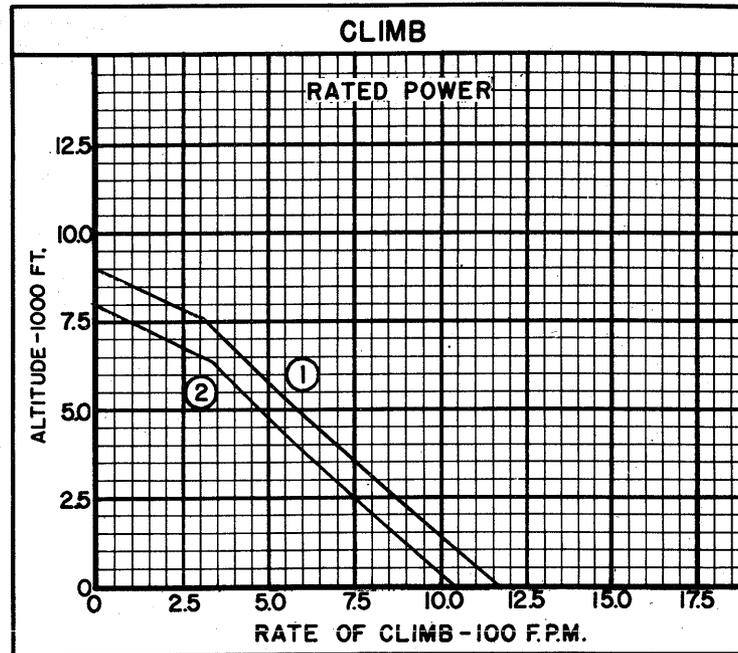
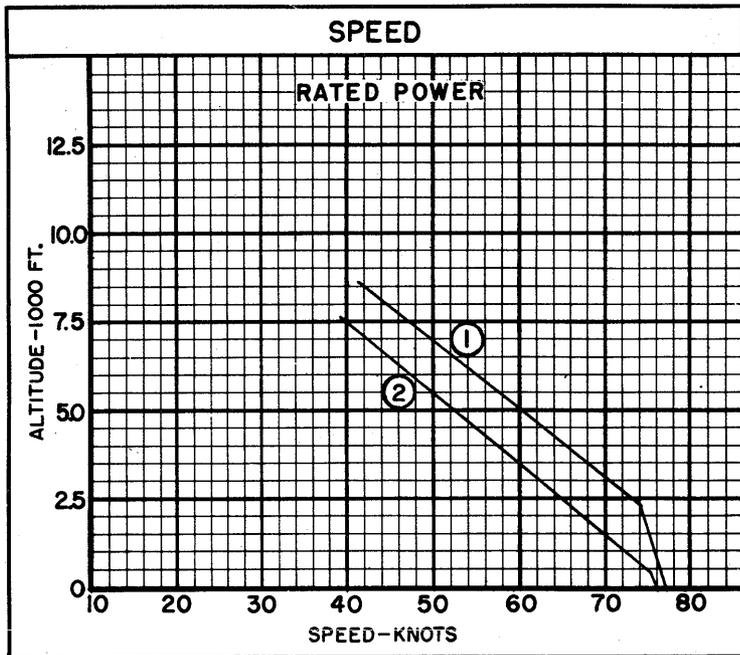
Warm-up and Take-off: 1 min at NRP at Sea Level  
 Cruise out: at Vmax at Sea Level  
 Hover: 12 min at Sea Level  
 Drop Weapon: No fuel consumed, no distance gained  
 Cruise back: at Vmax at Sea Level  
 Reserve: 10% of initial fuel load

MAX. RANGE MISSION

Warm-up and Take-off: 1 min at NRP at Sea Level  
 Cruise: at speed for best range at Sea Level  
 Reserve: 10% of initial fuel load

MAX. ENDURANCE MISSION

Warm-up and Take-off: 1 min at NRP at Sea Level  
 Cruise: at speed for best endurance at Sea Level  
 Reserve: 10% of initial fuel load

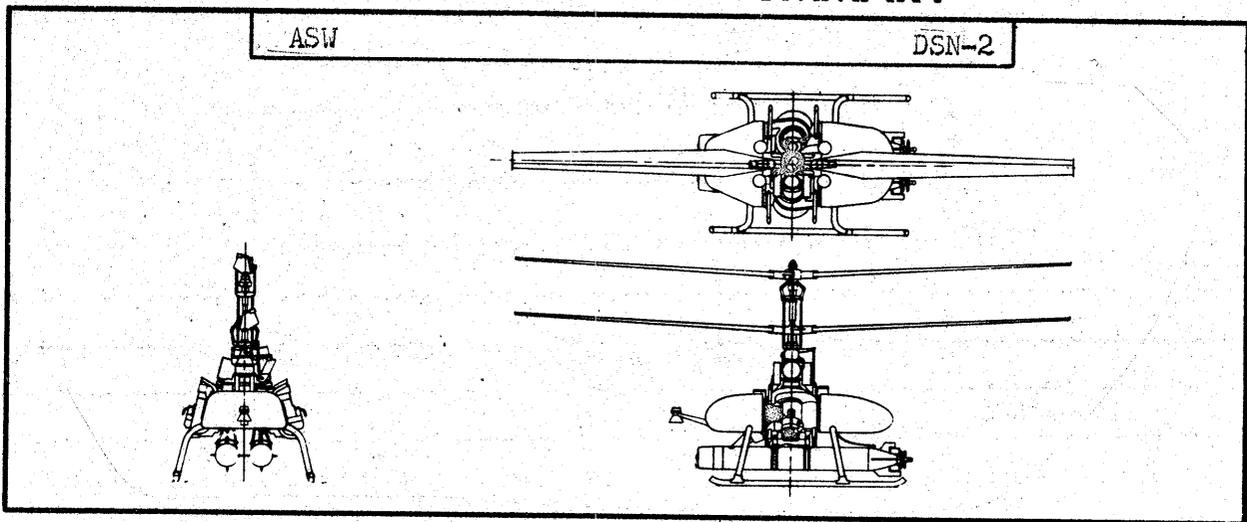


○ LOADING CONDITION COLUMN NUMBER

Standard Aircraft Characteristics NAVAER 1335E (Rev. 1-55)

# CHARACTERISTICS SUMMARY

ASW DSN-2



DISC AREA 314.2 sq. ft.  
 ROTOR DIA. 20' 0"

LENGTH 20' 0"  
 HEIGHT 7' 4"

AVAILABILITY			PROCUREMENT				
NUMBER AVAILABLE			NUMBER DELIVERED				
ACTIVE	RESERVE	TOTAL	IN FISCAL YEARS				

**STATUS**

First Flight ..... July 1960

**ENGINES**

Gyrodyne - Porsche  
 (2) YO-95-4

RATINGS

BHP/RPM/ALT

T.O. 72/4500/SSL  
 NORM. 72/4500/SSL

**FEATURES**

Completely ground controllable, including weapon drop

Flight path radar monitored

Designed to operate from destroyer-type vessels

**ARMAMENT**

Torpedoes:

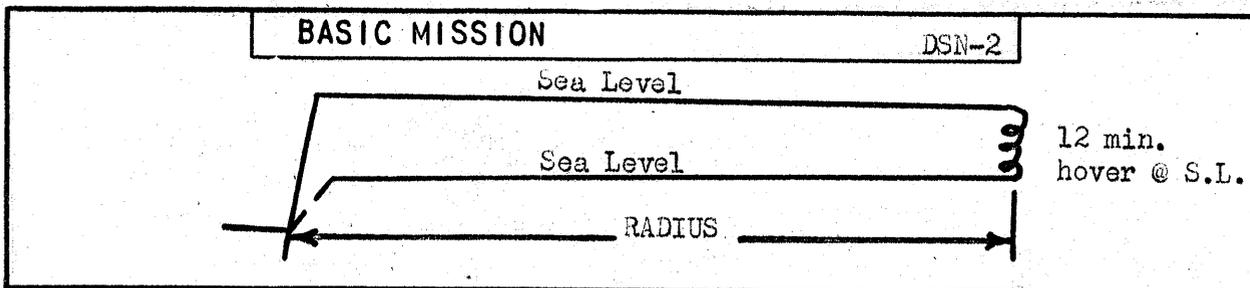
500 lb. .... 1

or

MK 43 ..... 2

NAVAER-1519E (Rev. 6-56)

# CHARACTERISTICS SUMMARY



PERFORMANCE		
ENDURANCE	RADIUS	SPEED
1.65 hours	32.0 naut. mi.	76.8 knots at Sea Level
42.5 knots avg.	77.5 knots avg.	60.5 knots at 5000 ft.
Sea Level	Sea Level	Normal Gross Weight Normal Power
1.04 hrs. mission time		
FORWARD FLIGHT CLIMB	SERVICE CEILING	HOVERING CEILING
1160 ft./min.	8600 ft.	2950 ft.
Sea Level, N. G. Wt., Normal Power	100 ft./min., N. G. Wt., Normal Power	N. G. Wt., Normal Power out of ground effect
		ft. N. G. Wt., Power in ground effect
LOAD	WEIGHTS	VERTICAL CLIMB
Fuel 95 lbs.	Empty 813 lbs.	640 ft./min.
Internal 95 lbs.	Normal Gross 1450 lbs.	Sea Level, N. G. Wt., Normal Power
External -- lbs.	Overload 1500 lbs.	
Payload 500 lbs.		

NOTES
1. Performance is based on calculations, modified by contractor flight test data of the Model XRON-1 Rotorcycle (17 ft. dia.)
2. Endurance and radius are based on manufacturer's fuel consumption data increased 5%.
3. Radius is computed at maximum speed outgoing and returning

NAVAER-1519D (Rev. 6-56)