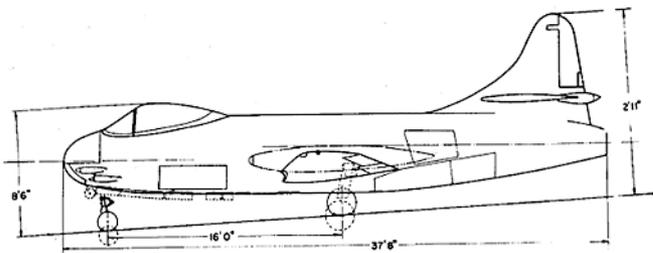
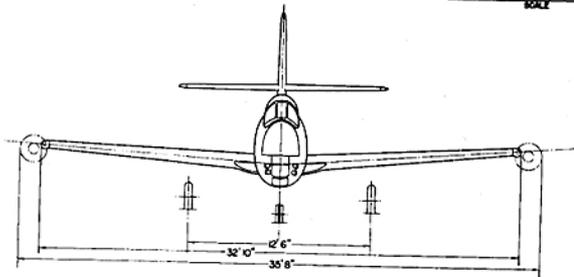
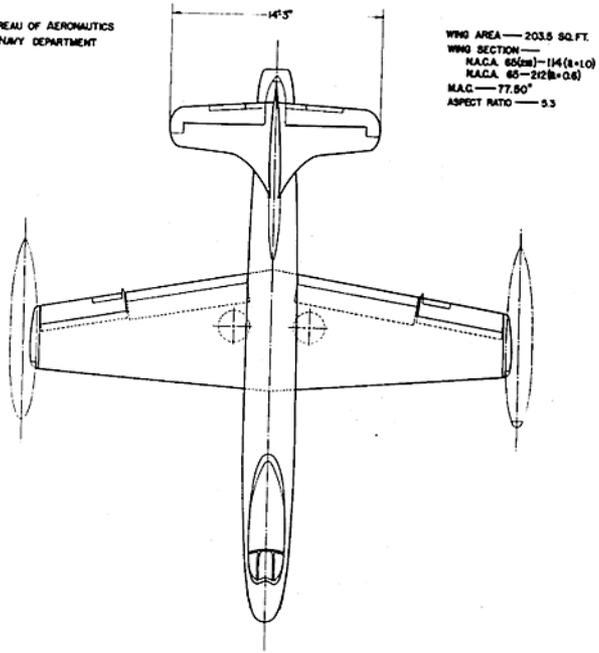


STANDARD AIRCRAFT CHARACTERISTICS
F6U-1 "PIRATE"

CHANCE VOUGHT

BUREAU OF AERONAUTICS
NAVY DEPARTMENT



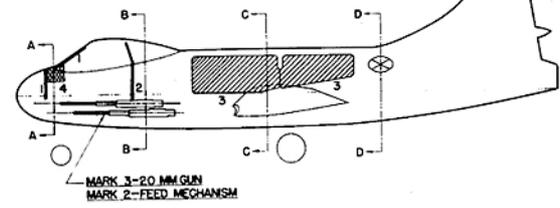
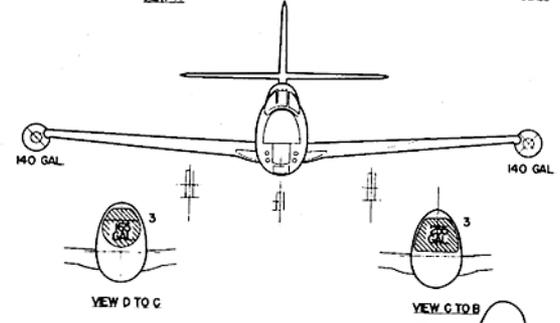
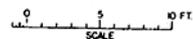
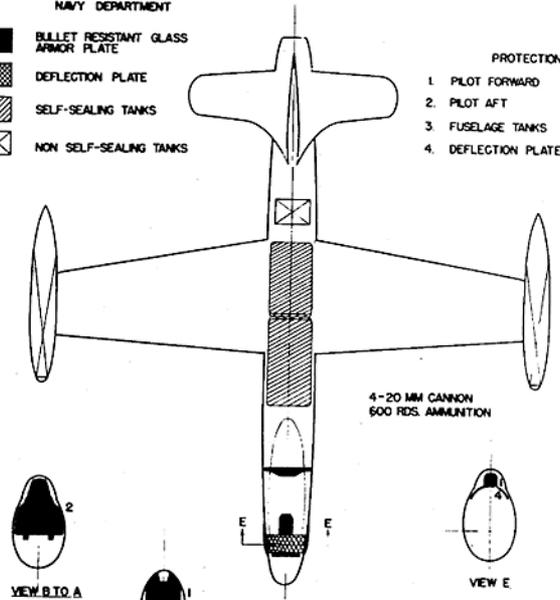
DESCRIPTIVE ARRANGEMENT

BUREAU OF AERONAUTICS
NAVY DEPARTMENT

- BULLET RESISTANT GLASS ARMOR PLATE
- DEFLECTION PLATE
- SELF-SEALING TANKS
- NON SELF-SEALING TANKS

PROTECTION

1. PILOT FORWARD	52.1 LBS
2. PILOT AFT	64.7 LBS
3. FUSELAGE TANKS	276.6 LBS
4. DEFLECTION PLATE	8.2 LBS



ARMAMENT & TANKS

MISSION AND DESCRIPTION

The F6U-1 is a general-purpose fighter, intended for use in escort, combat air patrol, and ground support missions, and also for fighter sweeps.

It is a single-seat, carrier-based aircraft. Fuselage is of conventional structure, with stainless steel construction around afterburner and tail-pipe. Cabin is pressurized. Wing structure employs metalite skin (dural-balsa-dural sandwich) approximately 1/4" thick, over conventional spar and rib structure.

Tail surfaces are conventional except for sandwich-type skin on fin. Slotted extensible wing flaps are fitted, and fuselage speed retarding brakes are provided.

DIMENSIONS

SPAN.....32'-10"
 LENGTH.....37'-8"
 HEIGHT.....12'-11"
 TREAD.....12'-6"
 WING AREA.....204 sq. ft.
 M.A.C.....77.5"

WEIGHTS

Loading	Lbs.	L.F.
EMPTY.....	7,320.....	
BASIC.....	7,931.....	
DESIGN.....	10,500.....	7.5
COMBAT.....	11,060.....	7.0
MAX.T.O....*	12,900.....	6.0
MAX.LAND....	9,300.....	

*Limited by structure in catapulting.

All weights calculated.

FUEL AND OIL

Gals.	No. Tanks	Location
420	2	Fuse., Seal.
280	2	Wing Tips

FUEL GRADE...115/145

FUEL SPEC...AN-F-48

OIL

CAPACITY (Gals.).....5
 GRADE.....1010
 SPEC.....AN-0-9

ELECTRONICS

VHF TRANSMITTER.....AN/ARC-1
 NAVIGATION.....AN/ARR-2A
 AUTO.D.F.....AN/ARN-6
 IFF.....AN/APX-1A
 RADIO ALTIMETER.....AN/APN-1

POWER PLANT

NO. & MODEL.....(1) J34-WE-30A
 MFR.....Westinghouse
 AFTERBURNER.....A-103B
 A.B.MFR.....Solar

RATINGS

Lbs. @ Rpm @ Alt.

T.O.(AB)	4,100	12,500	SSL
MIL.	3,150	12,500	SSL
COMB.(AB)	4,100	12,500	SSL
NORMAL	2,640	11,800	SSL

SPEC. NO. WAGT-24C4C-2

ORDNANCEGUNS

No.	Size	Location	Rds.
4	20 mm	Nose	600

(Space provision for 800 rds.)

FIRE CONTROL

Aircraft Fire
 Control System..Mk. 6, Mod.1



PERFORMANCE SUMMARY			
LOADING CONDITION		(1) Fighter 2-140 Gal. Tip Tanks	
TAKE-OFF WEIGHT	lbs.	12,874	
Fuel Fixed/Drop	lbs.	2,520/1,680	
Bombs	lbs.	None	
Wing/Power Loading (A)	lbs/sq.ft; lbs/bhp.	62.3/-	
Stall Speed--Power off	kn.	104.7	
Stall Speed--Power off - No Fuel	kn.	86.0	
Stall Speed--Power on	kn.	98	
Maximum Speed/Alt (B)	kn/ft.	400/18,000	
Take-off Distance, deck -- calm	ft.	1,805(2,905)	
Take-off Distance, deck 25 kn.	ft.	1,035(1,670)	
Take-off Distance, Airport	ft.		
Rate of climb -- sea level (B)	ft/min.	1,855	
Service Ceiling (B)	ft.	30,000	
Time-to-climb 20,000 ft. (B)	min.	17.2	
Time-to-climb 30,000 ft. (B)	min.	50.2	
Combat Range/V avClimbCruise	ft. n.mi/kn.	1,015/375	
Combat Radius/V av	ft. n.mi/kn.	390/375	
LOADING CONDITION		(2) Combat	(3) Combat
GROSS WEIGHT	lbs.	11,060	11,060
Engine power		Combat	Normal
Fuel	lbs.	2,520	2,520
Bombs/Tanks		None	None
Max. speed at sea level	kn.	518	395
Max. speed/Alt	kn/ft.	518/S.L.	429/22,000
Combat speed/Alt	kn/ft.	478/31,000	423/31,000
Rate of climb SL	ft/min.	8,060	2,385
Ceiling for 500 fpm R/C	ft.	46,300	29,000
Time-to-climb/Alt.	min/ft.	5.4/30,000	24.9/30,000

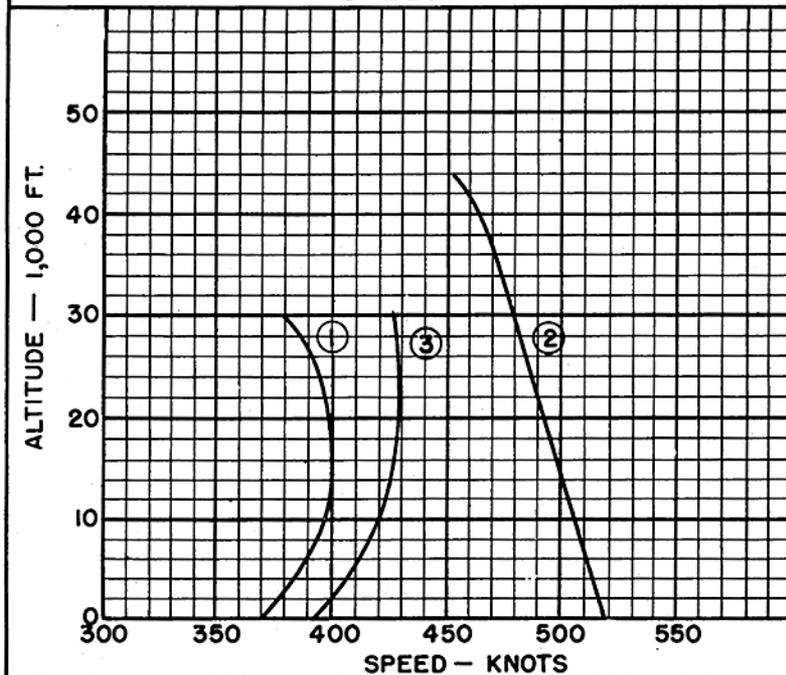
NOTES

- (A) BHP at Maximum Critical Altitude
- (B) Normal BHP

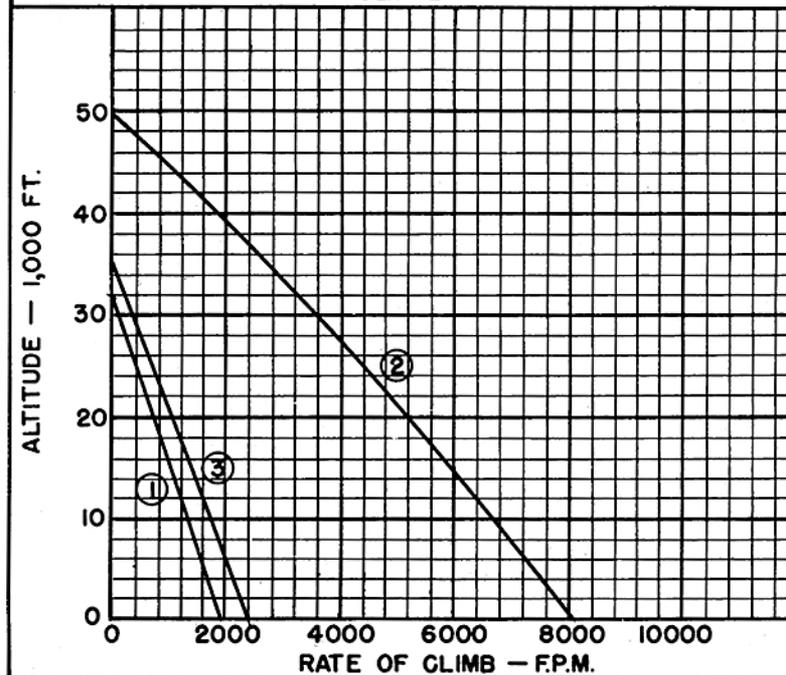
Performance is based on calculations. Range and radius are based on engine specification fuel consumption increased by 5%.

Normal take-off is by catapulting. Take-off figures in parentheses are for military power (afterburner inoperative); others are with take-off power (afterburner operating).

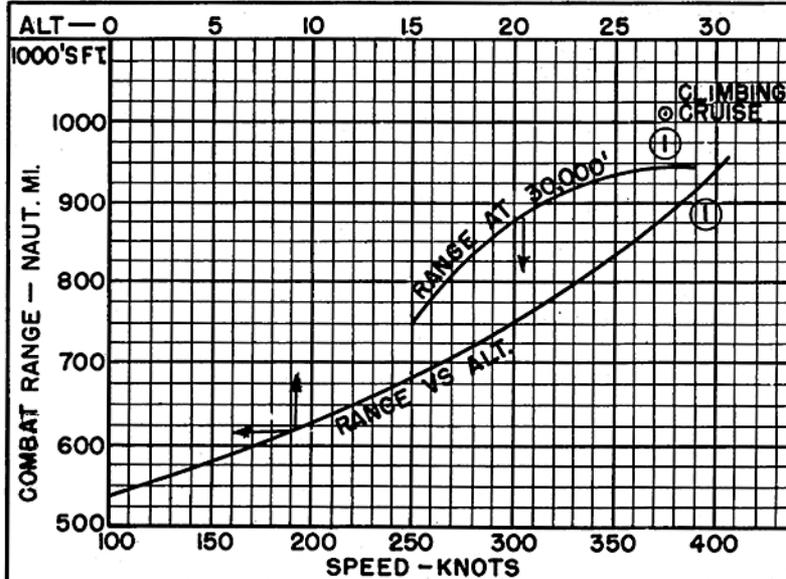
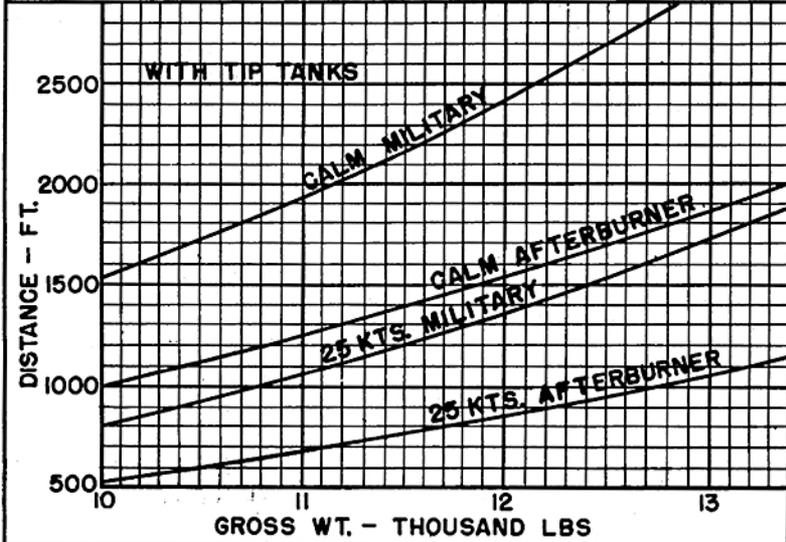
SPEED



CLIMB



TAKE-OFF



○ LOADING CONDITION COLUMN NUMBER

NOTES

Spotting: 200 ft. length is required to spot 38 airplanes of the 96 ft. wide deck immediately aft of the forward ramp on CV-9 class carriers.

ESCORT FIGHTER COMBAT RADIUS PROBLEM NO. F-5 (Gas Turbine Engine)

<u>WARM-UP TAXI TAKE-OFF</u>	<u>CLIMB (A)</u>	<u>CRUISE-OUT</u>	<u>DESCEND</u>	<u>COMBAT</u>	<u>CLIMB (B)</u>	<u>CRUISE-BACK</u>	<u>RESERVE</u>
5 min. at sea level static normal power of all engines	at max. rate with mil. power to initial cruise-out alt. (Alt. not greater than alt. for 300 ft./min. max. rate of climb with normal power)	with optimum range operation. (State altitudes and any special engine operations involved.)	to 35,000 ft. unless alt. at end of cruise-out is less (No fuel used, no distance made good) <u>DROP TANKS</u> only when empty and state when dropped	35,000 ft. or at alt. at end of cruise-out if less, and V_{max} . 15 min. at Mil. power of which 5 min. is with augmentation if available (Fuel used, but no distance made good)	to initial cruise-back alt. under same conditions as for Climb (A) (Fuel used and distance made good)	under same conditions as Cruise-Out	10% of total initial fuel load.

COMBAT RADIUS = CLIMB (A) / CRUISE-OUT = CRUISE-BACK / CLIMB (B)

Combat range is flown in climbing cruise. Starting altitude 26,400 feet; tanks dropped at 28,000 feet, 302 n. mi. from take-off. Climb to 30,400 feet. End cruise at 36,000 feet.

Combat radius altitudes are as below:

