

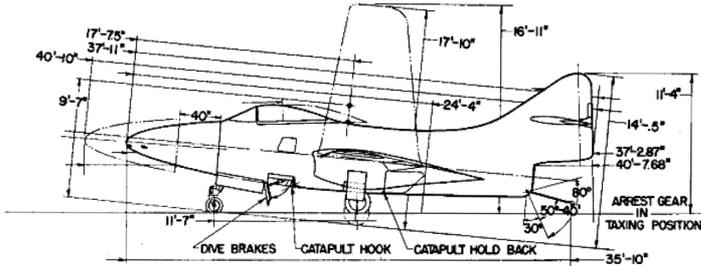
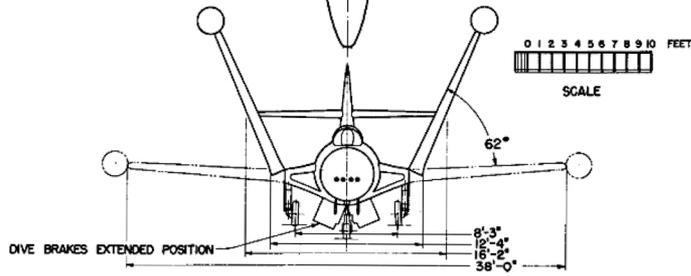
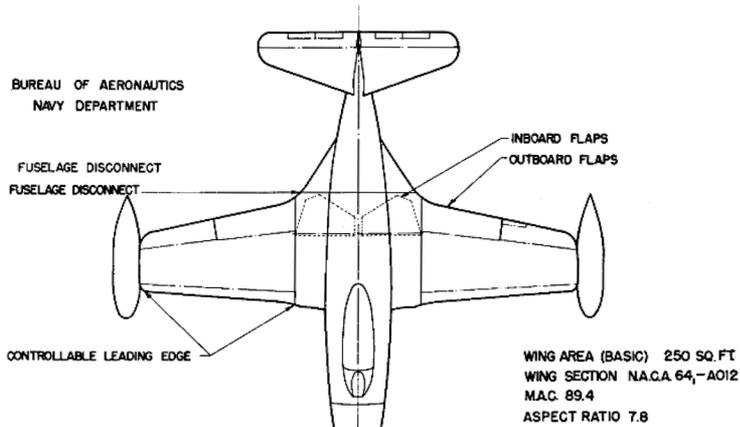
STANDARD AIRCRAFT CHARACTERISTICS

F9F-2 "PANTHER"

GRUMMAN

Standard Aircraft Characteristics Manual 1954 (REV. 1-49)

BUREAU OF AERONAUTICS
NAVY DEPARTMENT

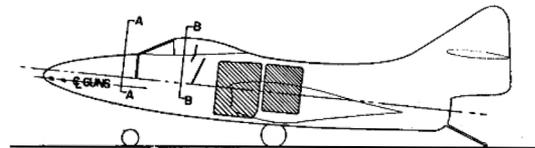
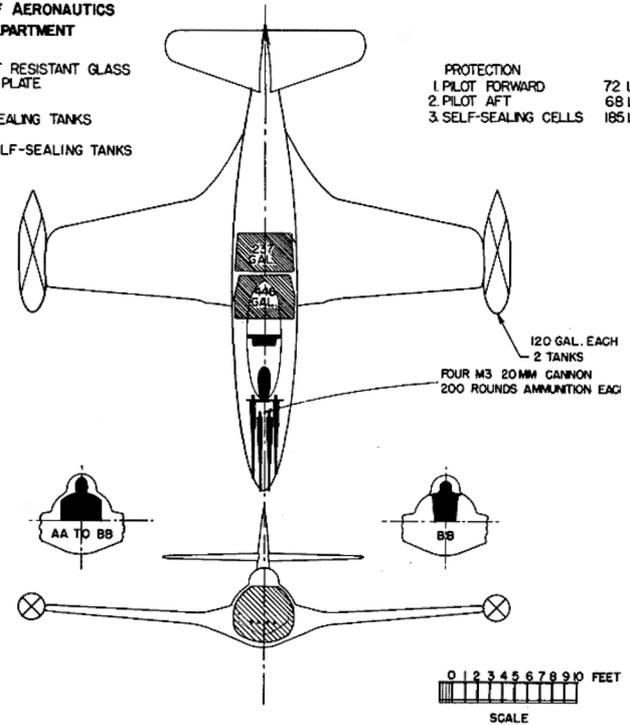


DESCRIPTIVE ARRANGEMENT

BUREAU OF AERONAUTICS
NAVY DEPARTMENT

- BULLET RESISTANT GLASS ARMOR PLATE
- ▨ SELF-SEALING TANKS
- ⊗ NON SELF-SEALING TANKS

- PROTECTION
- 1. PILOT FORWARD 72 LBS.
 - 2. PILOT AFT 68 LBS.
 - 3. SELF-SEALING CELLS 165 LBS.



ARMAMENT & TANKS

POWER PLANT

NO. & MODEL.....(1) J42-P-3
 MFR.....Prett & Whitney
 ENG. LENGTH.....104"
 ENG. DIAMETER.....50"

RATINGS

	Lbs.	@ Rpm	@ Alt.
T. O.			
(Wet) 5,750	12,300	8,800	S.S.L.
T. O. 5,000	12,300	8,800	S.S.L.
MIL. 5,000	12,300	8,800	S.S.L.
NORM. 4,000	11,600	8,800	S.S.L.

SPEC. NO. N-1603

ORDNANCE**GUNS**

No.	Size	Location	Rls.
4	20 mm	Fuselage	800

FIRE CONTROL

AFCS.....Mk. 6, Mod. 0
 Radar Ranging Equip....
 AN/APG-30

BOMBS AND ROCKETS

Type	Size	Location	No.
RYAR	5"	Outer Wing	6
Bombs	250#	Outer Wing	6
Bombs	500#	Outer Wing	4
Bombs	1000#	Inner Wing	2

(On Mk. 51 racks)

Any combination of above not to exceed 2,500 pounds.

MISSION AND DESCRIPTION

The F9F-2 is a single seat carrier-based fighter airplane whose mission is the destruction of opposing aircraft and ground support. Water injection is available to aid in take-off. A droop nose is fitted for take-off and landings.

Guns and radio are accessible by sliding forward the movable nose. The engine is serviced or changed quickly by removal of the tail section of the fuselage.

All internal fuel is in two interconnected tanks and two removable tip tanks also feed automatically into the main fuel tank.

The pressurized cabin has temperature control and a Grumman ejection seat.

For normal aileron control a hydraulic boost system is provided. In case of hydraulic failure a mechanical boost is available. The canopy is hydraulically operated.

Dive brakes are located under the fuselage. All control surfaces are metal covered and spot welded. All control surfaces are equipped with conventional trim tabs.

DIMENSIONS

WING AREA.....	250 sq. ft.
SPAN.....	38' - 0"
LENGTH.....	37' - 3"
HEIGHT.....	11' - 4"
TREAD.....	8' - 3"
M.A.C.....	7' - 5"

WEIGHTS

Loadings	Lbs.	L.F.
EMPTY.....	9,303
BASIC.....	10,107
DESIGN.....	13,000	7.0
COMBAT.....	14,235	7.0
MAX.T.O..(Field).....	19,494	6.0
MAX.LAND.(Field).....	14,200

All weights are actual.

* Maximum anticipated loading

FUEL AND OIL

Gal.	No. Tanks	Location
683	2	Fuselage (S.S.)
240	2	Wing, Tip

FUEL GRADE.....100/130
 FUEL SPEC.....MIL-F-5572

OIL

CAPACITY (Gals.).....2.2
 GRADE.....1010
 SPEC.....MIL-C-6081

ELECTRONICS

VEP COMM.....AN/ARC-1,-1A
 HOMING.....AN/ARR-2A
 ANF.....AN/ANF-5
 IFF.....AN/APX-6
 RADIO ALTIMETER.....AN/APX-1
 UHF D.F.....AN/ARA-25
 (Service Installation)
 UHF.....AN/ARC-27
 (Replacement for AN/ARC-1)
 HOMING.....AN/ARR-21
 (Replacement for AN/ARR-2A
 and AN/ARR-5)

PERFORMANCE SUMMARY

TAKE-OFF LOADING CONDITION	(1) FIGHTER 2 - 120 Gal. Tip Tanks	(3) GRD. SUPPORT 2-1,000# Bombs 6-5# HYAR Rock. Tip Tanks		
TAKE-OFF WEIGHT	lb.	16,450	19,494	
Fuel (Internal/Fixed Tip)	lb.	4,098/1,440	4,098/1,440	
Payload (Ammunition, Bombs, Rockets)	lb.	450	3,306	
Wing loading	lb./sq.ft.	65.8	78	
Stall speed - power-off	kn.	112	122	
Take-off run at S.L. - calm	ft.	(Dry) 2,330	(Dry) 3,450	
Take-off run at S.L. 25 km. wind	ft.	(Dry) 1,450	(Dry) 2,253	
Take-off to clear 50 ft. - calm	ft.	--	--	
Max. speed/altitude (1)	kn./ft.	457/22,000	342/S.L.	
Rate of climb at S.L. (2)	fpm	5,140	3,000	
Time: S.L. to 20,000 ft. (2)	min.	5.3	10.7	
Time: S.L. to 30,000 ft. (2)	min.	10.3	--	
Service ceiling (100 fpm) (2)	ft.	44,600	27,600	
Combat range	n.mi.	1,175	485	
Average cruising speed	kn.	423	290	
Cruising altitude(s)	ft.	40,800/43,000	25,000/20,000	
Combat radius	n.mi.	465	150	
Average cruising speed	kn.	435	294	
COMBAT LOADING CONDITION	(2) Tip Tanks	(4) 2-Mc 51 Rockets 6-Mc 9 Launchers		
COMBAT WEIGHT	lb.	14,235	14,423	
Engine power		Military	Military	
Fuel	lb.	3,323	3,323	
Combat speed/combat altitude	kn./ft.	460/35,000	459/S.L.	
Rate of climb/combat altitude	fpm/ft.	1,360/35,000	5,600/S.L.	
Combat ceiling (500 fpm)	ft.	43,400	40,000	
Rate of climb at S.L.	fpm	6,000	5,600	
Max. speed at S.L.	kn.	500	459	
Max. speed/altitude	kn./ft.	500/S.L.	462/15,000	
LANDING WEIGHT	lb.	11,477		
Fuel	lb.	1,036		
Stall speed - power-off	kn.	94.3		
Stall speed - with approach power	kn.	91.3		

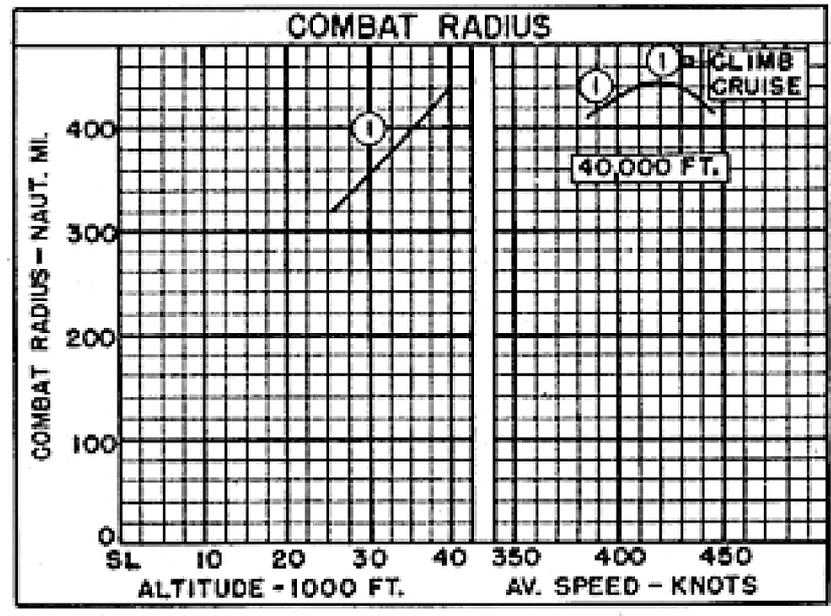
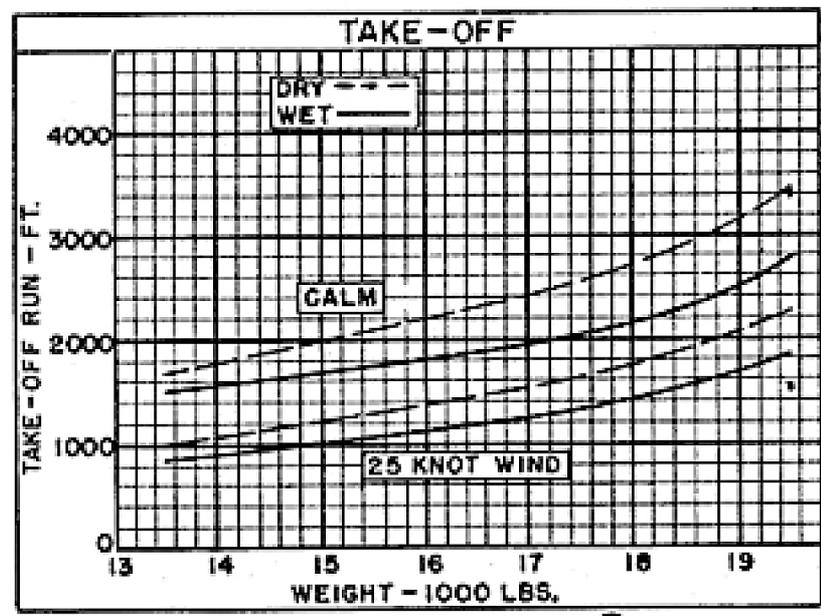
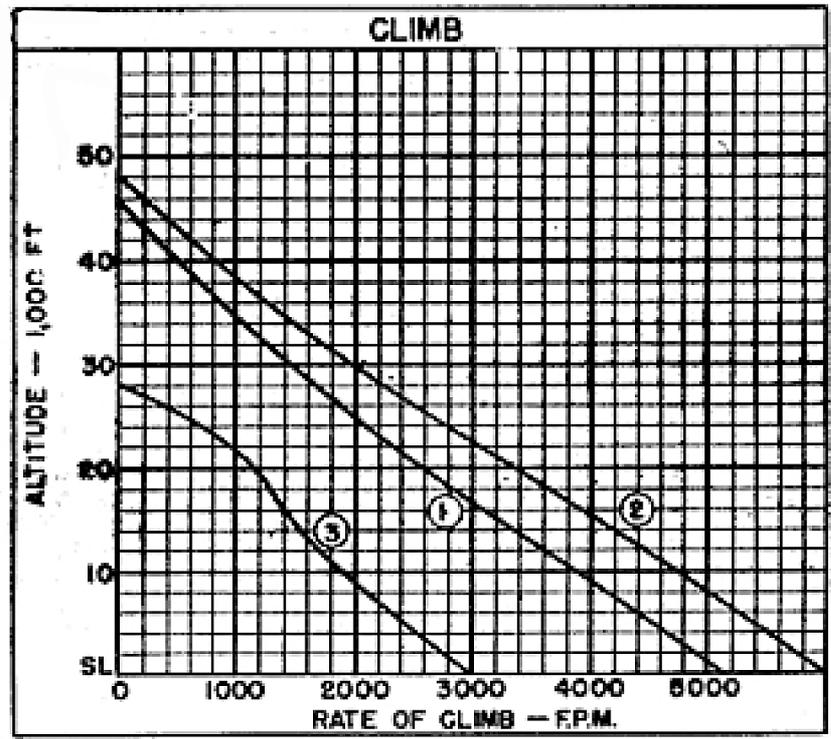
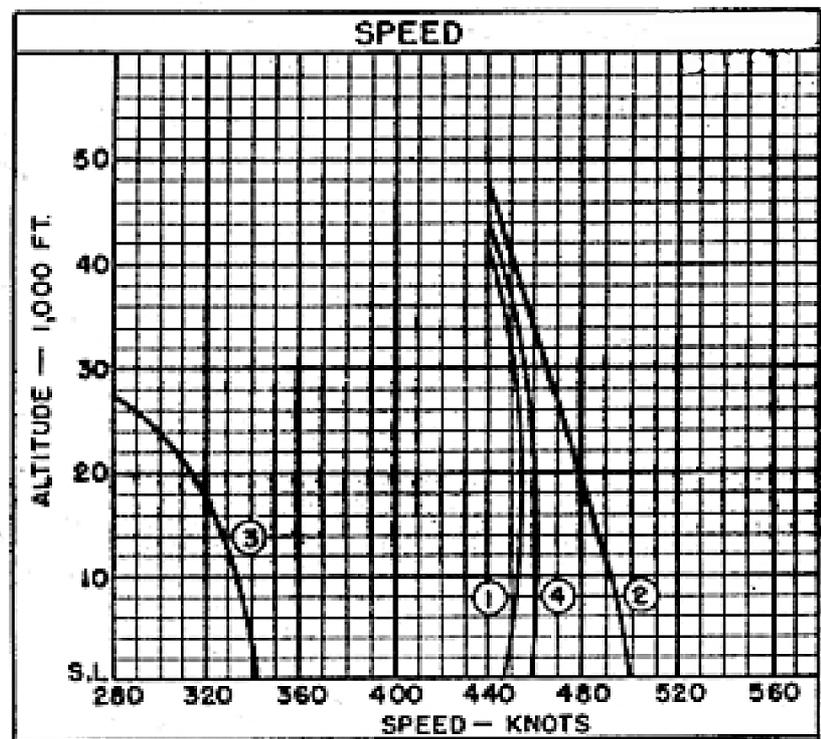
NOTES

- (1) Normal Power
(2) Military Power

Performance is based on NATC flight test of the F9F-2 airplane.

Range and radius are based on flight test fuel consumption increased by 5%.

Spotting: 200 ft. length is required to spot 28 airplanes on the 96 ft. wide deck immediately aft of the forward ramp on CV-9 class carriers.



Standard Aircraft Characteristics Number 1391 (REV. 2-50)

○ LOADING CONDITION COLUMN NUMBER

American Aviation Historical Society Archives - www.aaahs-online.org

NOTES

GENERAL PURPOSE AND ESCORT FIGHTER COMBAT RADIUS PROBLEM (GAS TURBINE)

WARM-UP, TAXI, TAKE-OFF: 5 minutes at normal power

CLIMB: To cruising ceiling at military power. (Cruising ceiling = altitude for 300 ft./min. at normal power)

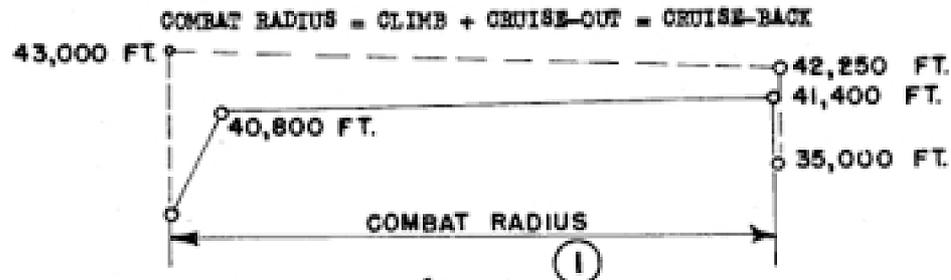
CRUISE-OUT: At V for long range at cruising ceiling

DESCEND: To 35,000 ft. (No fuel used, no distance gained)

COMBAT: At 35,000 ft. for 20 minutes at military power. (Assume combat concluded at initial cruise-back altitude)

CRUISE-BACK: At V for long range at cruising ceiling

RESERVE: 20 minutes at V for maximum endurance at Sea Level plus 5% of initial fuel load



Based on F-5 problem, combat radius would increase to 560 n.mi.

Based on reserve fuel allowance of F-5 problem (10% of initial fuel load), range would increase to 1,335 n.mi.

Radius is reduced approximately 6.5 n.mi. for each additional minute of combat.

GROUND SUPPORT FIGHTER COMBAT RADIUS PROBLEM (GAS TURBINE)

WARM-UP, TAXI, TAKE-OFF: 5 minutes at normal power

CLIMB: To cruising ceiling at military power. (Cruising ceiling = altitude for 300 ft./min. at normal power)

CRUISE-OUT: At V for long range at cruising ceiling

DESCEND: To sea level (No fuel used, no distance gained)

LOITER: 10 minutes at airspeeds for maximum endurance at sea level

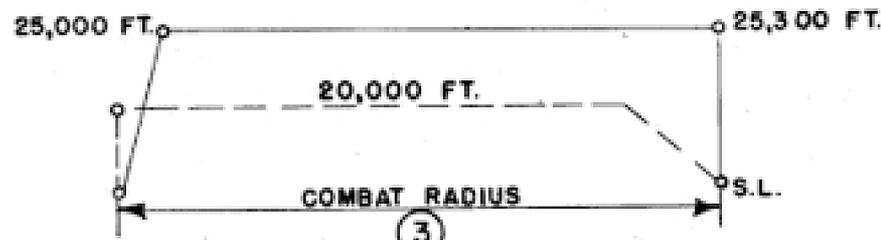
DROP BOMBS AND FIRE EXTERNAL ROCKETS

COMBAT: At sea level for 10 minutes at military power

CLIMB: To altitude for maximum radius (20,000 feet) at military power

CRUISE-BACK: At V for long range at 20,000 feet

RESERVE: 20 minutes at V for maximum endurance at Sea Level plus 5% of initial fuel load



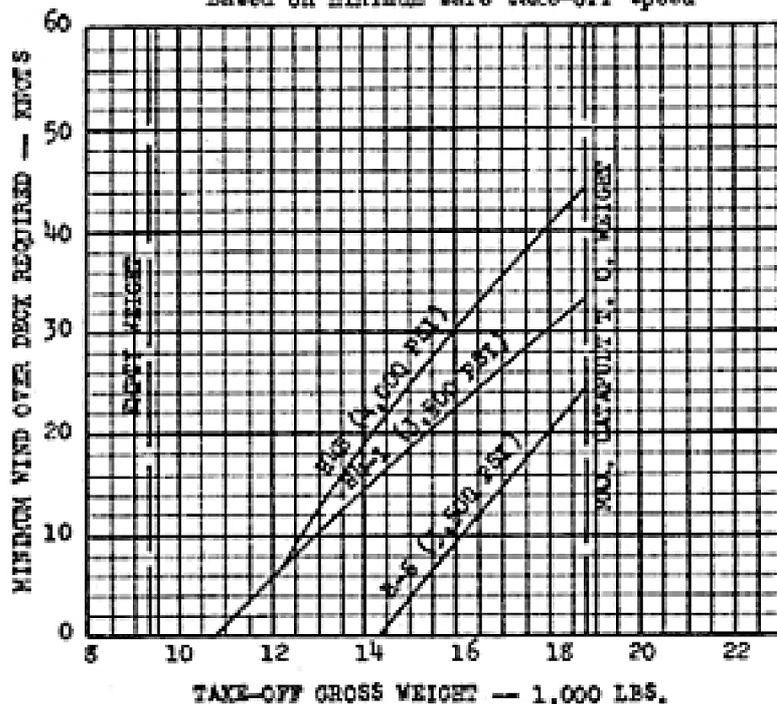
This chart supersedes previously issued chart dated 1 July 1949. Reason for reissue: Flight test data available.

CARRIER SUITABILITY

NOTE: These curves should be used for planning purposes only. Actual catapult and arresting gear operation should be in accordance with applicable Aircraft Technical Orders, and Catapult and Arresting Gear Bulletins.

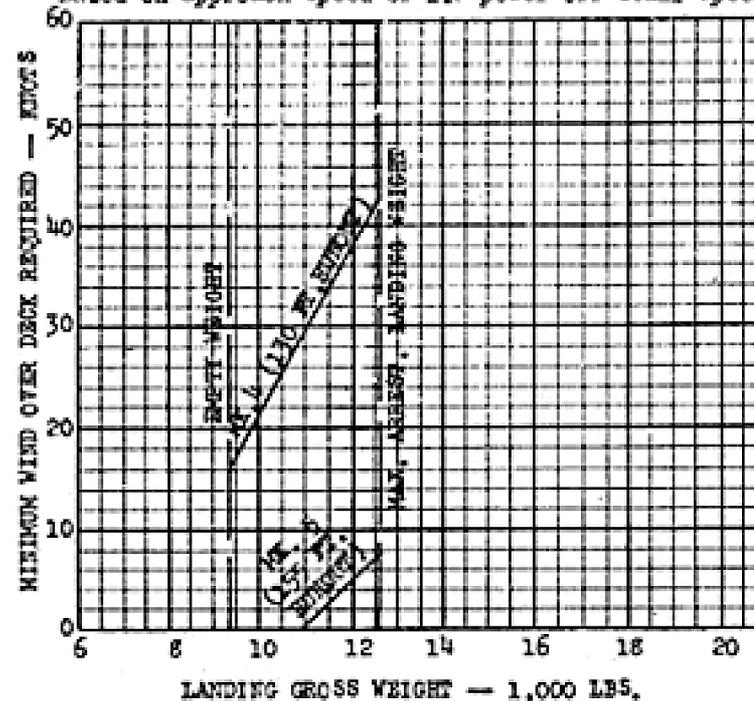
WIND OVER DECK REQUIRED FOR CATAPULTING VS. GROSS WEIGHT

Based on minimum safe take-off speed



WIND OVER DECK REQUIRED FOR LANDING VS. GROSS WEIGHT

Based on approach speed of 1.2 power-off stall speed



NOTE: No wind required for Mk. 7 arresting gear