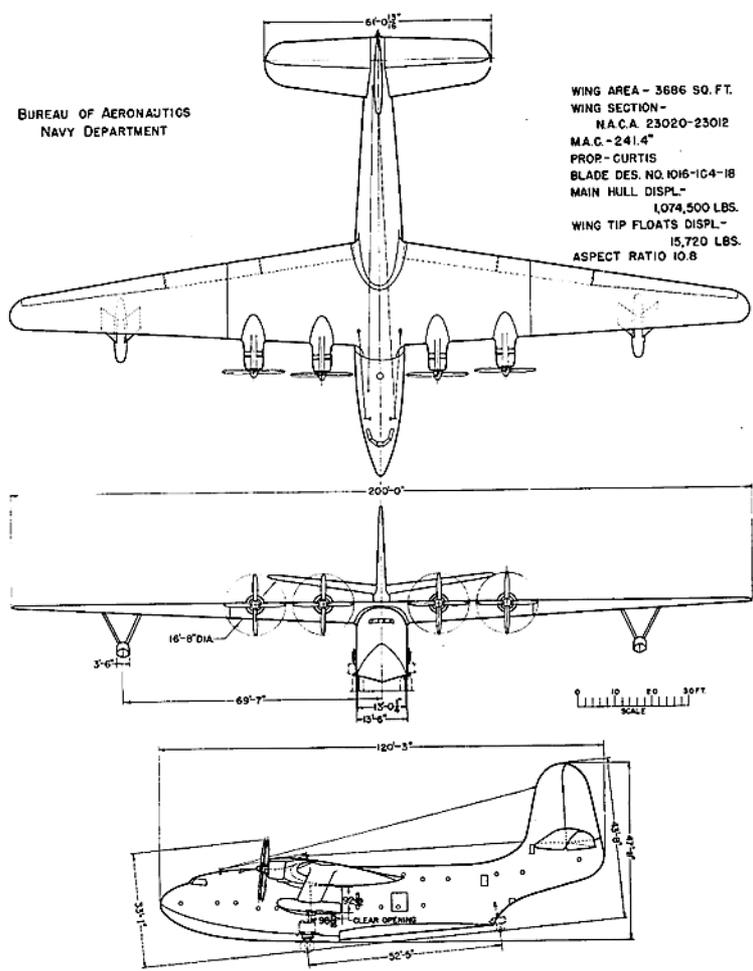


STANDARD AIRCRAFT CHARACTERISTICS
JRM-1 "MARS"

BUREAU OF AERONAUTICS
NAVY DEPARTMENT

WING AREA - 3686 SQ. FT.
WING SECTION -
N.A.C.A. 23020-23012
M.A.C. - 241.4"
PROP - CURTIS
BLADE DES. NO. 1016-1C4-18
MAIN HULL DISPL. -
1,074,500 LBS.
WING TIP FLOATS DISPL. -
15,720 LBS.
ASPECT RATIO 10.8



DESCRIPTIVE ARRANGEMENT

MISSION AND DESCRIPTION

Primary mission is for cargo and personnel transport. It is also designed for employment as troop (138 fully equipped men) and for casualty transport (100 litters plus ten attendants). The airplane takes off and lands on water and can be handled ashore using beaching gear.

This airplane is a development of the Model XPB2M-1R airplane (MARS) which was originally designed in 1935. It is a high-wing, all-metal, four-engine seaplane of conventional design. Numerous large hatches and hull doors with powered cargo hoist facilitate loading. Tie down fittings are provided in cargo compartments. Built for the U. S. Navy by Glenn L. Martin Co., to meet Navy and CAA transport specifications. Elevator and rudder control systems have boost assist. Inboard engines have reversible pitch propellers. Has jato-assisted take-off.

Rearward-moving slotted flaps fitted.

DIMENSIONS

SPAN.....200'-0"
 LENGTH.....120'-3"
 HEIGHT.....43'-8"
 WING AREA.....3685 sq. ft.
 M.A.C.....20'-1"

*Height is for airplane on beaching gear.

WEIGHTS

Loadings	Lbs.	L.F.
EMPTY.....	77609.....	
BASIC.....	80229.....	
DESIGN.....	165000.....	2.5
MAX. T.O.....	155000*.....	2.7
MAX. LD.....	155000.....	

*Limited by 3-Engine climb after T.O.
 Max. Wt. on Beaching Gear.....100,000
 All weights are actual.

FUEL AND OIL

Gal. - No. Tanks - Location
1220.....2.....Wing
12000.....6.....Hull
FUEL GRADE.....100/130
FUEL SPEC.AN-F-48

OIL

CAPACITY (gal.).....	616
SPEC.....	AN-O-8
GRADE.....	1120

ELECTRONICS

VHF COMMAND.....	AN/ARC-1
LOGAN.....	AN/APN-4
COMPASS.....	SCR-269
LP-MF-HF LIAISON.....	AN/ART-13 and AN/ARC-5
P/A SYSTEM.....	
MARKER BEACON REC.....	AN/ARN-8
ALTIMETER.....	AN/APN-1
IFF.....	AN/APX-2

POWER PLANT

NO. & MODEL.....	(4) R-3350-8
MFR.....	Wright
SUPERCH.....	1 Stage, 2 Speed
PROP. GEAR RATIO.....	16:7
PROP MFR.	Curtiss
PROP DESIGN NO.....	1016-1C4-18
NO. BL./DIA.....	4/16'-8"

RATINGS

	Bhp. @ Rpm.	@ Alt.
T.O.	2400 2600	S.L.
Mil.	2250 2600	4700
	1900 2600	15800
Norm.	2100 2400	4000
	1800 2400	16400

SEE NOTE

SPEC NO. N-779-F

ACCOMMODATIONS

CREW.....	11
TROOPS.....	138
LITTERS.....	100 (plus 10 seats)
PASSENGERS.....	63
CARGO SPACE.....	455 sq. ft.
FLOOR LOADING.....	300 psf

Miniature bridge crane under wing root with 5000# electric hoist.



PERFORMANCE SUMMARY				
LOADING CONDITION		(1) Cargo Overload	(2) Cargo Passenger	(3) Troop Transport 132 Troops
TAKE-OFF WEIGHT	lbs	155000	145000	145000
Fuel	lbs	49356	39810	26718
Bombs	lbs			
Payload	lbs	20000	20000	34000
Wing/Power Loading (A)	lbs/so.ft.lbs/bhp	42.1/21.5	39.4/20.1	39.4/20.1
Stall Speed--Power off	kn	79.4	76.8	76.8
Stall Speed--Power off - No Fuel	kn	65.5	65.4	69.4
Stall Speed--Power on	kn	69.2	62.0	62.0
Maximum Speed/Alt (B)	kn/ft	194/5800	197/5800	197/5800
Take-off Distance, deck -- calm	ft			
Take-off Distance, deck	kn. ft			
Take-off Time	Sec.	155	55	55
Rate of climb -- sea level (B)	ft/min	615	720	720
Service Ceiling (B)	ft	19100	20800	20800
Time-to-climb 10000 ft. (B)	min	20.5	16.6	16.6
Time-to-climb 20000 ft. (B)	min		49.0	49.0
Combat Range/V av 1500	ft. n.mi/kn	3075/126	2500/125	1600/126
Combat Radius/V av	ft. n.mi/kn			
LOADING CONDITION				
GROSS WEIGHT	lbs			
Engine power				
Fuel	lbs			
Bombs/Tanks				
Max. speed at sea level	kn			
Max. speed	ft. kn			
Combat speed/Alt.	kn/ft			
Rate of climb SL	ft/min			
Ceiling for 500 fpa R/C	ft			
Time-to-climb/Alt.	min/ft			

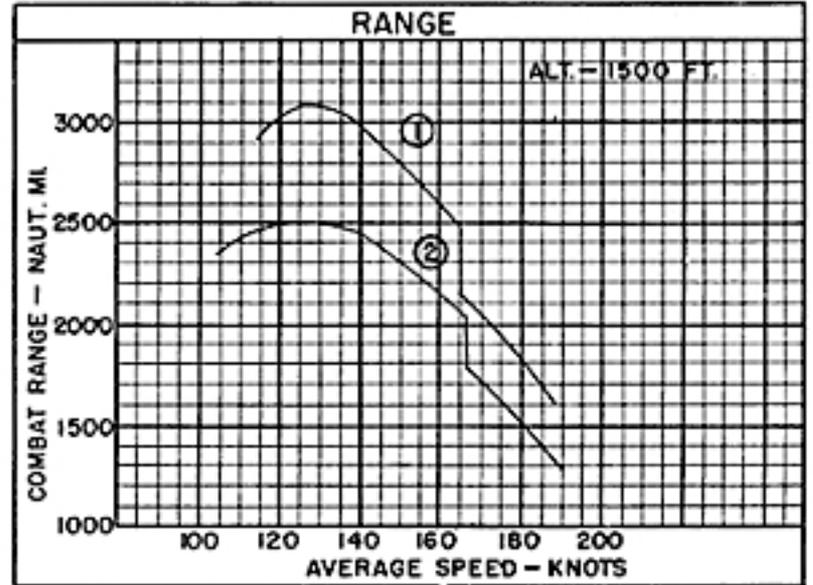
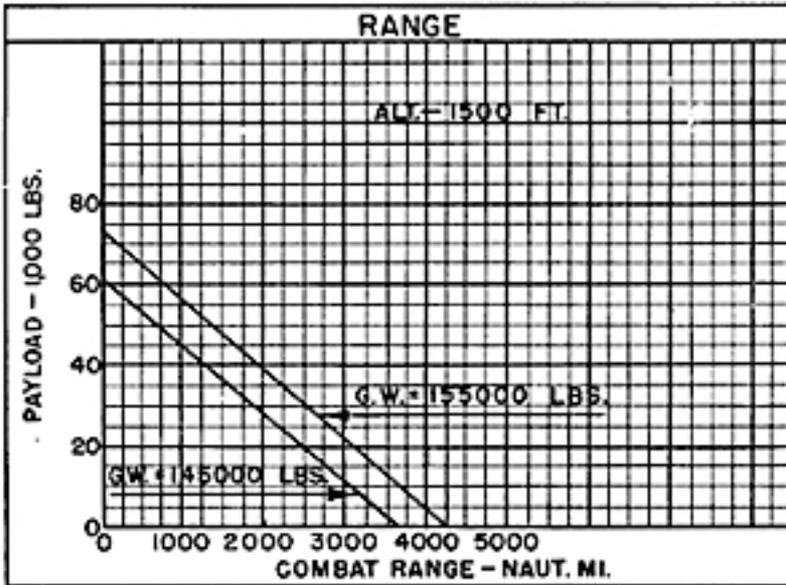
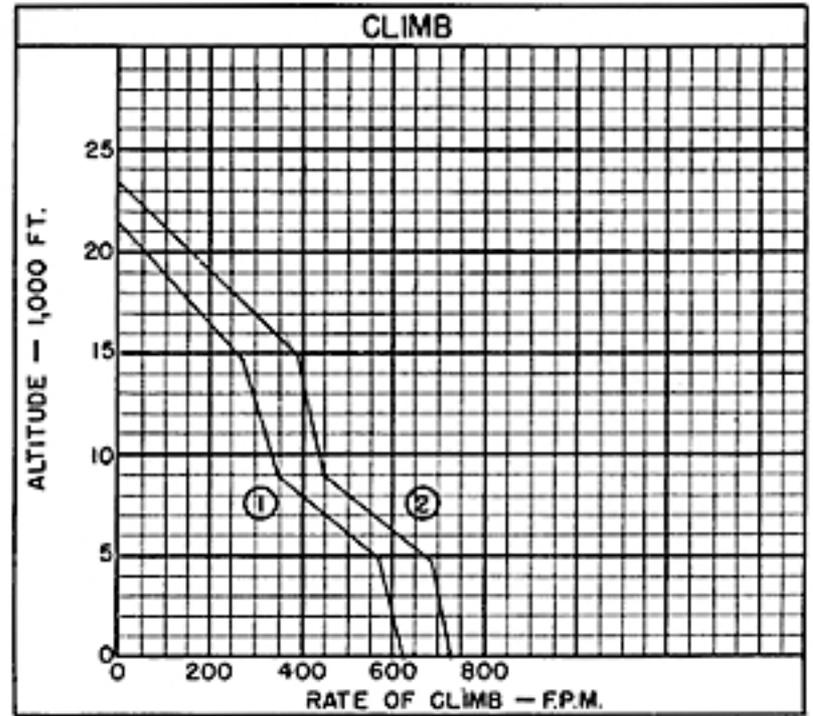
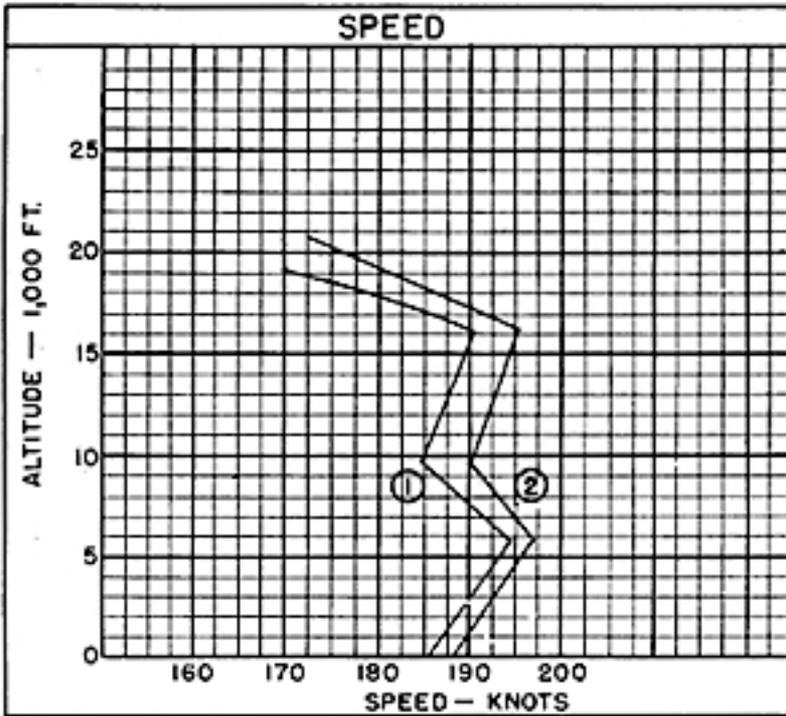
NOTES

- (A) HMF at Maximum Critical Altitude
(B) Normal HMF

Performance is based on flight test of the JRM-1 airplane. Range is based on flight test fuel consumption data of the JRM-1 increased by 5%.

Performance calculated without de-icing equipment. Addition of de-icing equipment reduces Vmax by 4 knots and maximum combat range at 1500 ft. by 2.5%.

NOTES CONTINUED ON LAST PAGE



○ LOADING CONDITION COLUMN NUMBER

NOTES

Performance with one engine inoperative, flaps neutral, and feathered propeller is estimated to be:

Gross Weight.....145,000#
 Rate of Climb-S.L. - T.O. Power.....510 ft/min.
 Service Ceiling - Normal Power.....9,000 ft.

 Take-off time in parenthesis is for JATO, using twelve 14-AS-1000-D4 units fired in banks of four.

Engine ratings from Flight Test:

	<u>Eng.</u>	<u>Eng.</u>	<u>Alt.</u>
T.O.	2400	2600	S.L.
Norm.	2100	2400	S.L. to 4800
	1800	2400	9000' to 14300